

 **BOEING**  
COMPONENT  
MAINTENANCE MANUAL

TO: ALL HOLDERS OF AILERON CONTROL COMPONENTS COMPONENT MAINTENANCE MANUAL  
27-11-10

REVISION NO. 2 DATED JUL 01/99

HIGHLIGHTS

Pages which have been added or revised are outlined below together with the highlights of the revision. Remove and insert the affected pages as listed and enter Revision No. and date on the Record of Revision Sheet.

CHAPTER/SECTION

AND PAGE NO.

DESCRIPTION OF CHANGE

TITLE PAGE

Added 251T1512-6 fitting assembly.

1

REPAIR 4-1

601-604

1002,1004-1005,1007,

1009,1011-1015

CONTENTS

1

Added check instruction to repair instruction for  
251T1512-5 fitting assembly.

REPAIR 3-1

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REPAIR 3-1

602-604

Edited illustration without technical change.

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HIGHLIGHTS

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## AILERON CONTROL COMPONENTS

PART NUMBERS 251T1289-1  
251T1509-1  
251T1512-5,-6

COMPONENT MAINTENANCE MANUAL  
WITH  
ILLUSTRATED PARTS LIST

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TITLE PAGE

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REVISION RECORD

- Retain this record in front of manual. On receipt of revision, insert revised pages in the manual, and enter revision number, date inserted and initial.

REVISION NUMBER	REVISION DATE	DATE FILED	BY	REVISION NUMBER	REVISION DATE	DATE FILED	BY

TEMPORARY REVISION AND SERVICE BULLETIN RECORD

BOEING SERVICE BULLETIN	BOEING TEMPORARY REVISION	OTHER DIRECTIVE	DATE OF INCORPORATION INTO MANUAL

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TR & SB RECORD

01

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PAGE	DATE	CODE	PAGE	DATE	CODE
27-11-10			REPAIR 3-1		CONT.
			*604	JUL 01/99	01.1
TITLE PAGE			REPAIR 4-1		
*1	JUL 01/99	01.1	*601	JUL 01/99	01.1
2	BLANK		*602	JUL 01/99	01.1
REVISION RECORD			*603	JUL 01/99	01.1
1	OCT 10/84	01	*604	JUL 01/99	01.1
2	BLANK		ILLUSTRATED PARTS LIST		
TR & SB RECORD			1001	OCT 10/84	01
1	OCT 10/84	01	*1002	JUL 01/99	01.1
2	BLANK		*1003	JUL 01/99	01.1
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602	OCT 10/84	01	*1015	JUL 01/99	01.1
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601	OCT 10/84	01			
602	OCT 10/84	01			
REPAIR 2-1					
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*601	JUL 01/99	01.1			
*602	JUL 01/99	01.1			
*603	JUL 01/99	01.1			

\* = REVISED, ADDED OR DELETED

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EFFECTIVE PAGES  
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01 Jul 01/99

TABLE OF CONTENTS

NOTE: This manual contains overhaul data for various components of the Aileron Control Installation. Functions which cannot be performed by the use of standard industry practices are included in the repair instructions for each component.

AILERON CONTROL COMPONENTS

<u>PART NUMBER</u>	<u>NOMENCLATURE</u>	<u>PAGE</u>
--	REPAIR-GENERAL	601, REPAIR-GEN
251T1289	LINK ASSY	601, REPAIR 1-1
251T1509	ROD ASSY	601, REPAIR 2-1
251T1512	FITTING ASSY	601, REPAIR 3-1 601, REPAIR 4-1
--	ILLUSTRATED PARTS LIST	1001

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### INTRODUCTION

The instructions in this manual provide the information necessary to perform maintenance functions ranging from simple checks and replacement to complete shop-type repair.

This manual is divided into separate sections:

- |                                                    |                              |
|----------------------------------------------------|------------------------------|
| 1. Title Page                                      | 4. List of Effective Pages   |
| 2. Record of Revisions                             | 5. Table of Contents         |
| 3. Temporary Revision &<br>Service Bulletin Record | 6. Introduction              |
|                                                    | 7. Procedures & IPL Sections |

Refer to the Table of Contents for the page location of applicable sections. An asterisked flagnote \*[ ] in place of the page number indicates that no special instructions are provided since the function can be performed using standard industry practices.

The beginning of the REPAIR section includes a list of the separate repairs, a list of applicable standard Boeing practices, and an explanation of the True Position Dimensioning symbols used.

An explanation of the use of the Illustrated Parts List is provided in the Introduction to that section.

All weights and measurements used in the manual are in English units, unless otherwise stated. When metric equivalents are given they will be in parentheses following the English units.

Design changes, optional parts, configuration differences and Service Bulletin modifications create alternate part numbers. These are identified in the Illustrated Parts List (IPL) by adding an alphabetical character to the basic item number. The resulting item number is called an alpha-variant. Throughout the manual, IPL basic item number references also apply to alpha-variants unless otherwise indicated.

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INTRODUCTION

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REPAIR – GENERAL

1. Content

- A. Each separate repair includes, as applicable, check, repair and refinish instructions.

2. Standard Practices

- A. Refer to the following standard practices as applicable, for details of procedures in individual repairs.

20-21-01 Magnetic Particle Inspection  
20-20-02 Penetrant Methods of Inspection  
20-30-02 Stripping of Protective Finish  
20-41-01 Decoding Table for Boeing Finish Codes  
20-41-02 Application of Chemical and Solvent Resistant Finishes  
20-42-03 Hard Chrome Plating  
20-43-01 Chromic Acid Anodizing  
20-50-03 Bearing Installation and Retention  
20-60-03 Lubricants

3. Materials

NOTE: Equivalent substitutes may be used.

- A. Primer -- BMS 10-11, Type 1 (Ref 20-60-02)  
B. Sealant -- BMS 5-95 (Ref 20-60-04)  
C. Enamel -- BMS 10-11, type 2, gloss gray (BAC 707) (Ref 20-60-02)  
D. Grease -- BMS 3-24 (Ref 20-60-03)

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REPAIR-GENERAL

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#### 4. Dimensioning Symbols

A. Standard True Position Dimensioning Symbols used in applicable repair procedures are shown in Fig. 601.

—	STRAIGHTNESS	$\oplus$	THEORETICAL EXACT POSITION OF A FEATURE (TRUE POSITION)
$\square$	FLATNESS	$\varnothing$	DIAMETER
$\perp$	PERPENDICULARITY (OR SQUARENESS)	BASIC (BSC) OR	A THEORETICALLY EXACT DIMENSION USED TO DESCRIBE SIZE, SHAPE OR LOCATION OF A FEATURE FROM WHICH PERMISSIBLE VARIATIONS ARE ESTABLISHED BY TOLERANCES ON OTHER DIMENSIONS OR NOTES.
//	PARALLELISM	<b>DIM</b>	
$\bigcirc$	ROUNDNESS	<b>-A-</b>	DATUM
$\bigcirc$	CYLINDRICITY	$\textcircled{M}$	MAXIMUM MATERIAL CONDITION (MMC)
$\frown$	PROFILE OF A LINE	$\textcircled{S}$	REGARDLESS OF FEATURE SIZE (RFS)
$\triangle$	PROFILE OF A SURFACE	$\textcircled{P}$	PROJECTED TOLERANCE ZONE
$\odot$	CONCENTRICITY		
$\equiv$	SYMMETRY		
$\sphericalangle$	ANGULARITY		
$\nearrow$	RUNOUT		

#### EXAMPLES

$\text{—} \quad 0.002$	STRAIGHT WITHIN 0.002	$\textcircled{\odot} \text{ C } \varnothing \quad 0.0005$	CONCENTRIC TO C WITHIN 0.0005 DIAMETER (FULL INDICATOR MOVEMENT)
$\perp \text{ B } \quad 0.002$	PERPENDICULAR TO B WITHIN 0.002	$\equiv \text{ A } \quad 0.010$	SYMMETRICAL WITH A WITHIN 0.010
$\parallel \text{ A } \quad 0.002$	PARALLEL TO A WITHIN 0.002	$\sphericalangle \text{ A } \quad 0.005$	ANGULAR TOLERANCE 0.005 WITH A
$\bigcirc \quad 0.002$	ROUND WITHIN 0.002	$\oplus \text{ B } \varnothing \quad 0.002 \textcircled{S}$	LOCATED AT TRUE POSITION WITHIN 0.002 DIA IN RELATION TO DATUM B, REGARDLESS OF FEATURE SIZE
$\bigcirc \quad 0.010$	CYLINDRICAL SURFACE MUST LIE BETWEEN TWO CONCENTRIC CYLINDERS, ONE OF WHICH HAS A RADIUS 0.010 INCH GREATER THAN THE OTHER	$\perp \text{ A } \varnothing \quad 0.010 \textcircled{M}$ $0.510 \textcircled{P}$	AXIS IS TOTALLY WITHIN A CYLINDER OF 0.010-INCH DIAMETER, PERPENDICULAR TO, AND EXTENDING 0.510-INCH ABOVE, DATUM A, MAXIMUM MATERIAL CONDITION
$\frown \text{ A } \quad 0.006$	EACH LINE ELEMENT OF THE SURFACE AT ANY CROSS SECTION MUST LIE BETWEEN TWO PROFILE BOUNDARIES 0.006 INCH APART IN RELATION TO DATUM PLANE A	$2.000$	EXACT DIMENSION IS 2.000
$\triangle \text{ A } \quad 0.020$	SURFACES MUST LIE WITHIN PARALLEL BOUNDARIES 0.02 INCH APART AND EQUALLY DISPOSED ABOUT TRUE PROFILE	OR $2.000$ BSC	

True Position Dimensioning Symbols  
Figure 601

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REPAIR-GENERAL

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LINK ASSEMBLY - REPAIR 1-1

251T1289-1

NOTE: Refer to REPAIR-GEN for list of applicable standard practices. For repair of surfaces which may only require restoration of original finish, refer to Refinish instructions, Fig. 601. Item numbers refer to IPL Fig. 1.

1. Penetrant check link (35) per 20-20-02.

2. Repair

A. Bearing Replacement (Fig. 601)

(1) Remove bearing (30).

(2) Install replacement bearing with wet BMS 5-95 sealant and roller or anvil swage bearing over housing per 20-50-03. After swaging, bearing must withstand a push-out load of 1095 pounds.

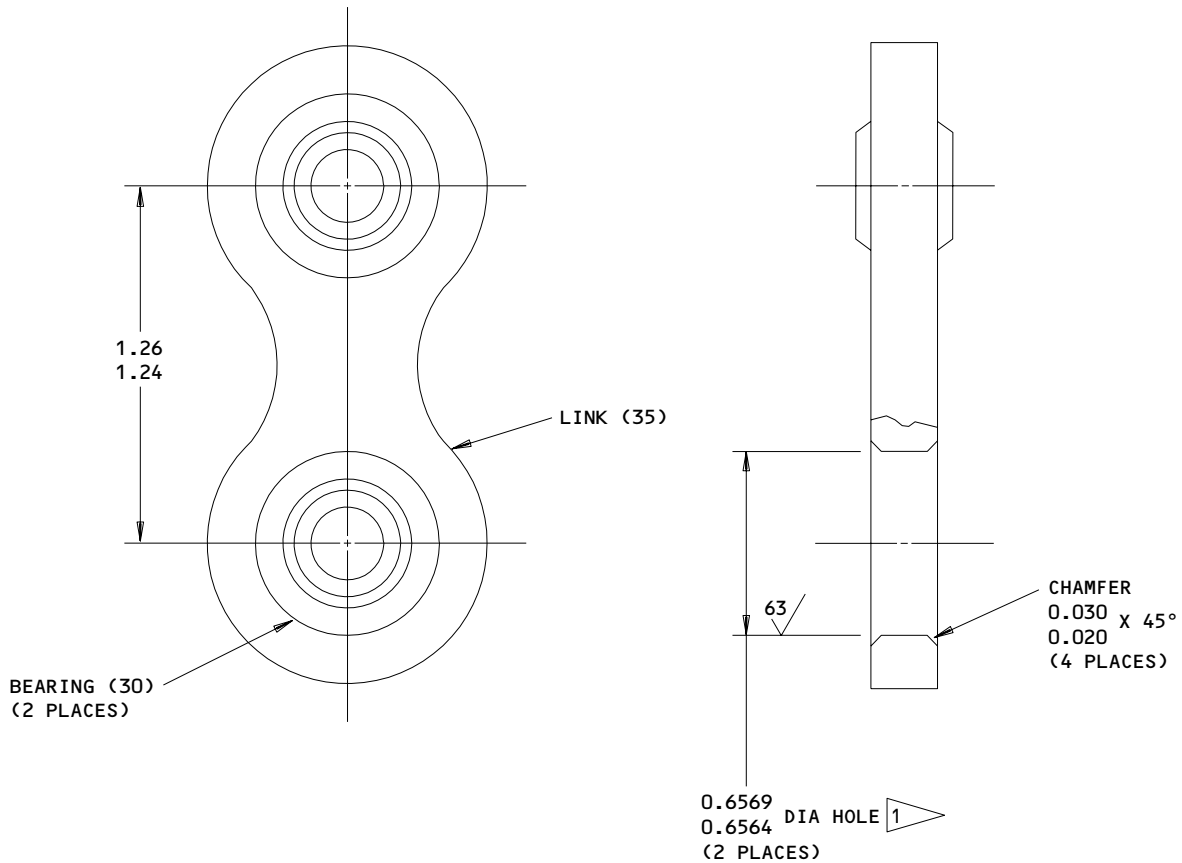
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REPAIR 1-1

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**REFINISH**

LINK (35) -- CHEMICAL TREAT OR CHROMIC ACID ANODIZE AND APPLY TWO COATS OF BMS 10-11, TYPE 1, PRIMER (F-18.03). APPLY ONE COAT BMS 10-11, TYPE II ENAMEL, COLOR 707 GRAY GLOSS (F-21.02).

MATERIAL: AL ALLOY

ALL DIMENSIONS ARE IN INCHES

1 OMIT PRIMER AND ENAMEL THIS SURFACE

ITEM NUMBERS REFER TO IPL FIG. 1

251T1289-1  
 Link Assembly Repair  
 Figure 601

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REPAIR 1-1

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ROD ASSEMBLY – REPAIR 2-1

251T1509-1

NOTE: Refer to REPAIR-GEN for list of applicable standard practices. For repair of surfaces which may only require restoration of original finish, refer to Refinish instructions, Fig. 601. Item numbers refer to IPL Fig. 2.

1. Magnetic particle inspect fitting (40, 45) and body (50) per 20-20-01.

2. Repair

A. Bearing Replacement (Fig. 601)

- (1) Remove rod bearing (35).
- (2) Position new rod bearing in fitting to dimension shown. Drill holes in new bearing using fitting as a guide.
- (3) Install bearing in fitting (40, 45) with BMS 5-95 sealant on faying surfaces.
- (4) Install rivets (30) with BMS 5-95 sealant.

B. Fitting Replacement (Fig. 601)

- (1) Remove end assembly (20, 25) from body (50) by loosening nut (15) and unscrewing.
- (2) Install end assembly (20, 25) using BMS 3-24 grease on both internal body threads and external fitting threads.
- (3) Adjust rod length to reference dimension shown and tighten nuts (15).

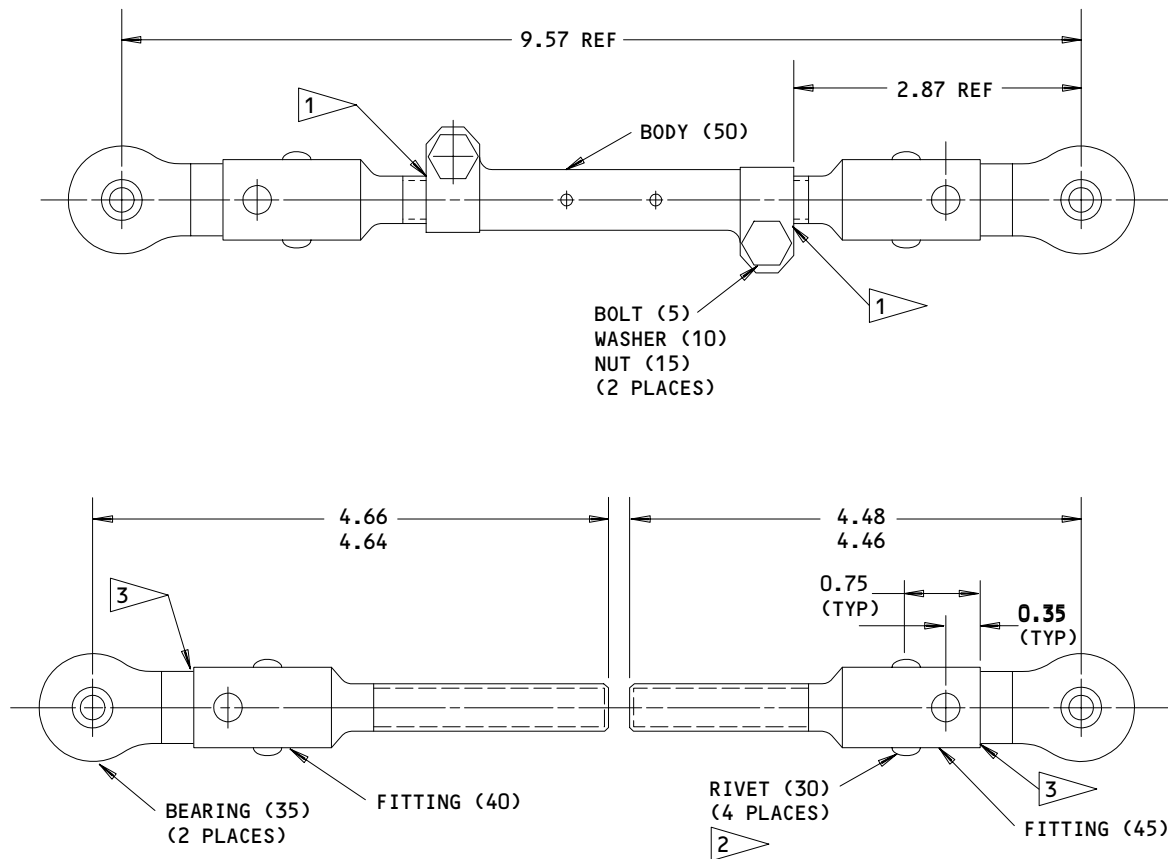
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REPAIR 2-1

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**REFINISH**

FITTING (40,45), BODY (50,50A) -- PASSIVATE  
 (F-17.09)

- 1 APPLY BMS 3-24 GREASE ON BOTH INTERNAL AND EXTERNAL THREADS
- 2 INSTALL RIVET (30) WITH BMS 5-95 SEALANT
- 3 ASSEMBLE WITH BMS 5-95 SEALANT ON FAYING SURFACES

MATERIAL: FITTING (40,45), BODY (50,50A) --  
 15-5PH CRES  
 150-170 KSI

ALL DIMENSIONS ARE IN INCHES

ITEM NUMBERS REFER TO IPL FIG. 2

251T1509-1  
 Rod Assembly Repair  
 Figure 601

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REPAIR 2-1

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FITTING ASSEMBLY – REPAIR 3-1

251T1512-5

NOTE: Refer to REPAIR-GEN for list of applicable standard practices. For repair of surfaces which may only require restoration of original finish, refer to Refinish instructions, Fig. 601. Item numbers refer to IPL Fig. 3.

Do not disassemble bracket (15), bolt (30), spacer (35), collar (50) or nutplates (25) unless necessary for repair or replacement.

1. Penetrant inspect fitting (70), bracket (15) and spacer (65) per 20-20-02.
2. Bushing Replacement (Fig. 601)
  - A. Remove bushing (55).
  - B. Install replacement bushing per 20-50-03 with BMS 3-24 grease.
3. Hole Repair (Fig. 601)
  - A. Machine hole oversize as required, within repair limit shown to remove defects.
  - B. Check the fitting (70) hole to the dimension shown in Fig. 601.
  - C. Manufacture repair bushing per Fig. 601.

NOTE: Repair bushing is special oversize bushing to replace installation part. Bushings are not part of assembly.

- D. Attach repair bushing to fitting assembly with aluminum wire and attach tag stating "Special Oversize Bushing Must Be Used."

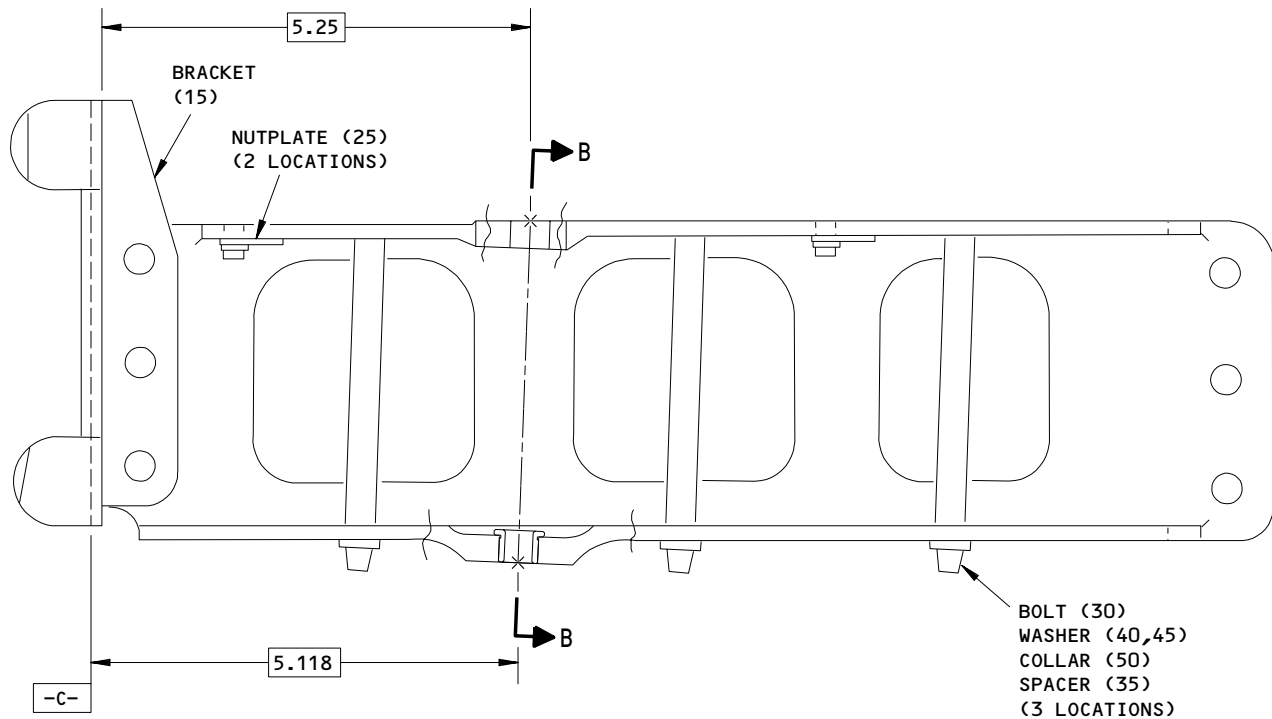
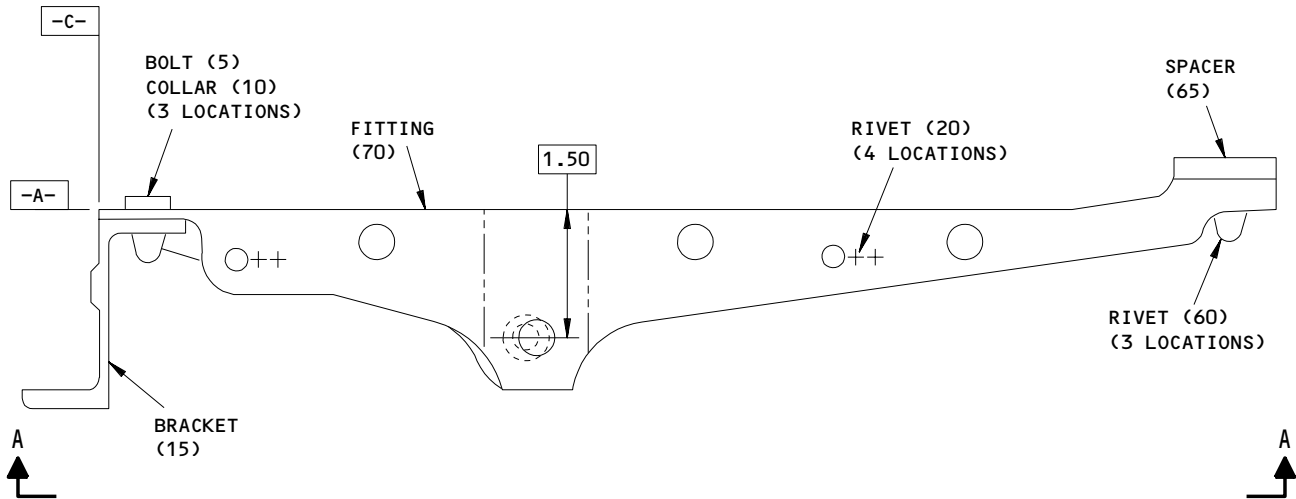
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A-A

251T1512-5  
 Fitting Assembly Repair  
 Figure 601 (Sheet 1)

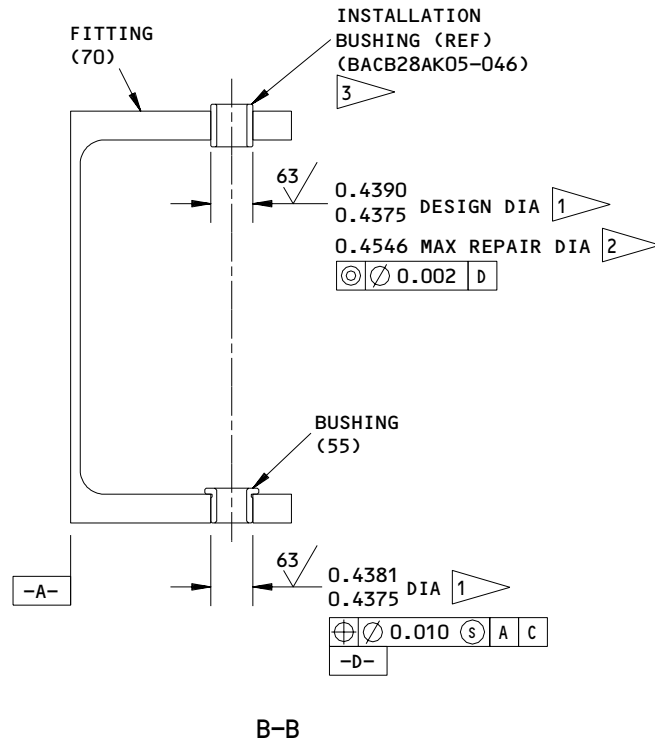
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**REFINISH**

FITTING (70), SPACER (65), BRACKET (15) --  
CHROMIC ACID ANODIZE AND APPLY ONE LAYER OF  
BMS 10-11, TYPE 1 PRIMER (F-18.13) UNLESS  
SHOWN DIFFERENTLY

**REPAIR**

REF 2 3

MATERIAL: FITTING (70), SPACER (65) AND  
BRACKET (15) - AL ALLOY

ITEM NUMBERS REFER TO IPL FIG. 3

ALL DIMENSIONS ARE IN INCHES

- 1 NO PRIMER THIS SURFACE
- 2 REPAIR LIMIT FOR INSTALLATION OF REPAIR BUSHING
- 3 OVERSIZE BUSHING REPLACES BACB28AK05-064 INSTALLATION PART. ATTACH TO ASSEMBLY. TAG ASSEMBLY TO USE OVERSIZE BUSHING DURING INSTALLATION

251T1512-5  
Fitting Assembly Repair  
Figure 601 (Sheet 2)

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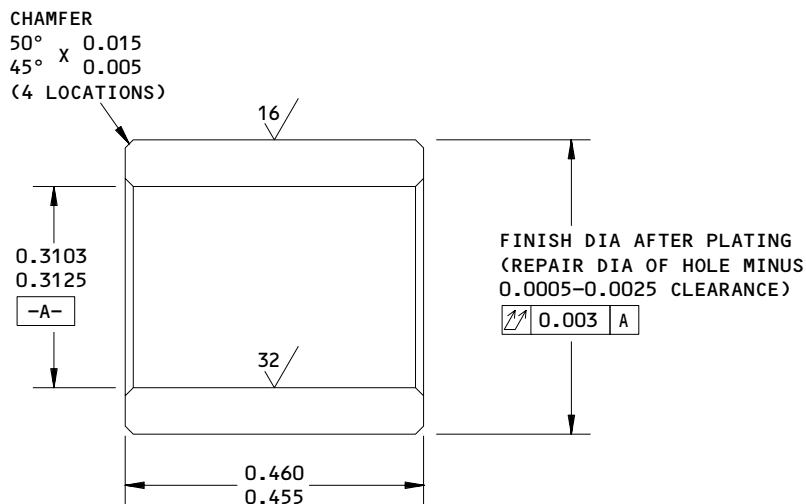
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OVERSIZE REPLACEMENT FOR BUSHING BACB28AK05-046

63/ ALL MACHINED SURFACES UNLESS SHOWN  
 DIFFERENTLY

MATERIAL: AL-NI-BRONZE REFER TO AMS 4880 OR  
 AMS 4640

BREAK SHARP EDGES 0.01-0.03 R

FINISH: CADMIUM PLATE 0.0003-0.0005 THICK  
 REFER TO SOPM 20-42-05

ALL DIMENSIONS ARE IN INCHES

Bushing Detail  
 Figure 602

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REPAIR 3-1

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FITTING ASSEMBLY – REPAIR 4-1

251T1512-6

NOTE: Refer to REPAIR-GEN for a list of applicable standard practices. For repair of surfaces which may only require restoration of original finish, refer to Refinish instructions, Fig. 601. Item numbers refer to IPL Fig. 6.

1. Penetrant inspect the fitting (30) per 20-20-02.
2. Bushing Replacement (Fig. 601)
  - A. Remove the bushing (25) from the fitting (30).
  - B. Install the replacement bushing per 20-50-03 with BMS 3-24 grease or BMS 5-95 sealant.
3. Hole Repair (Fig. 601)
  - A. Machine hole oversize as required, within the repair limit shown to remove defects.
  - B. Check the fitting (30) hole to the dimension shown in Fig. 601.
  - C. Manufacture repair bushing per Fig. 602.

NOTE: Repair bushing is special oversize bushing to replace installation part. Bushings are not part of assembly.
  - D. Attach the repair bushing to the fitting (30) assembly with aluminum wire and attach tag stating "Special Oversize Bushing Must be Used."

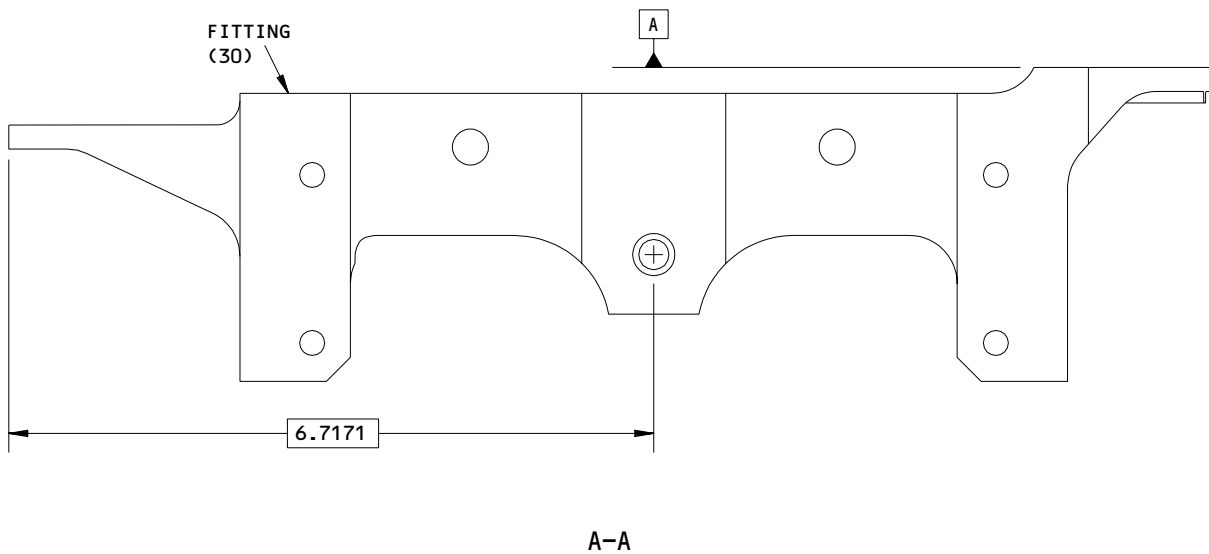
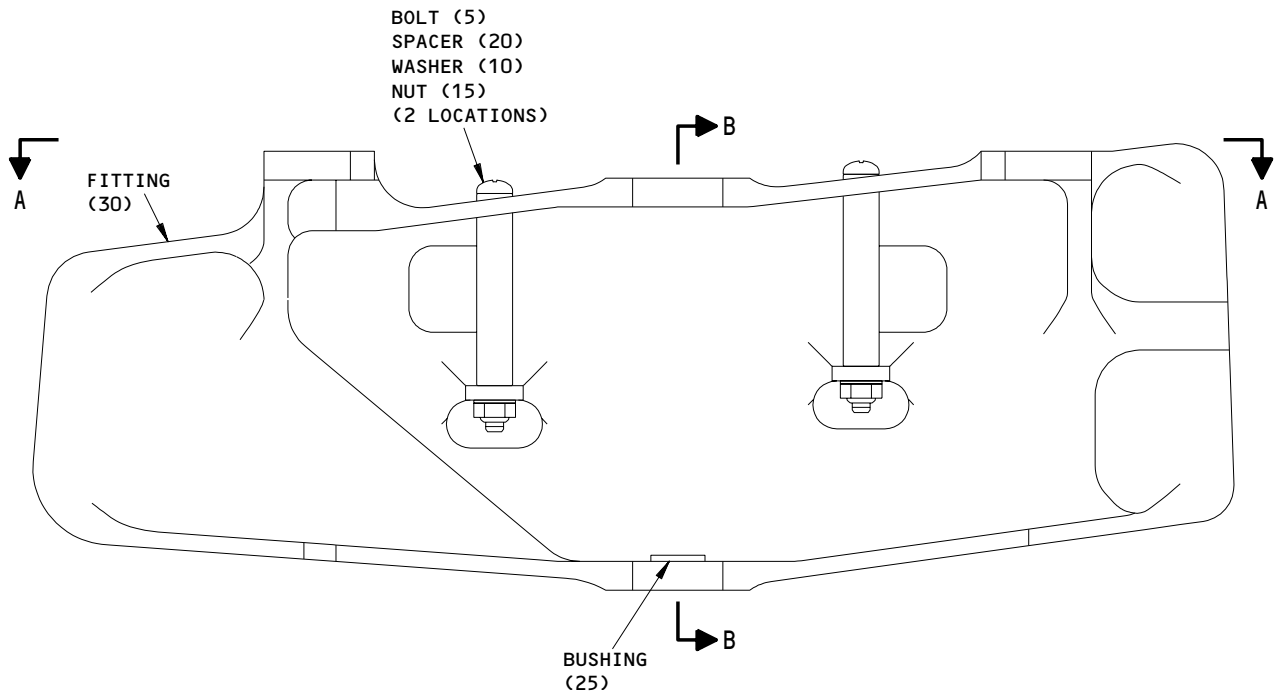
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251T1512-6  
 Fitting Assembly Repair  
 Figure 601 (Sheet 1)

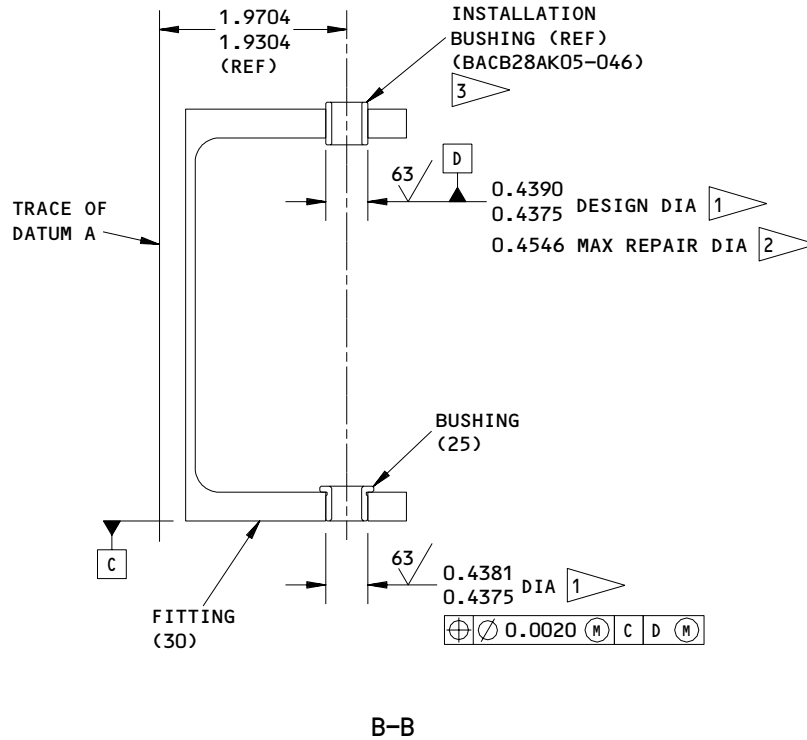
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**REFINISH**

FITTING (30) - BORIC ACID - SULFURIC ACID ANODIZE (F-17.35) AND A LAYER OF BMS 10-11, TYPE 1 PRIMER (F-20.03) UNLESS SHOWN DIFFERENTLY

**REPAIR**

REF 2 3

MATERIAL: FITTING (30) - AL ALLOY  
ITEM NUMBERS REFER TO IPL FIG. 6  
ALL DIMENSIONS ARE IN INCHES

- 1 NO PRIMER THIS SURFACE
- 2 REPAIR LIMIT FOR INSTALLATION OF REPAIR BUSHING
- 3 OVERSIZE BUSHING REPLACES BACB28AK05-064 INSTALLATION PART. ATTACH TO ASSEMBLY. TAG ASSEMBLY TO USE OVERSIZE BUSHING DURING INSTALLATION

251T1512-6  
Fitting Assembly Repair  
Figure 601 (Sheet 2)

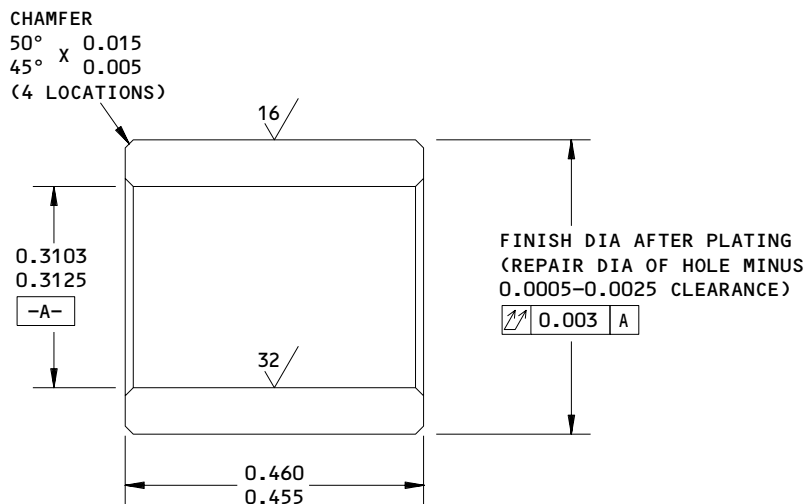
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OVERSIZE REPLACEMENT FOR BUSHING BACB28AK05-046

63/ ALL MACHINED SURFACES UNLESS SHOWN  
 DIFFERENTLY

MATERIAL: AL-NI-BRONZE REFER TO AMS 4880 OR  
 AMS 4640

BREAK SHARP EDGES 0.01-0.03 R

FINISH: CADMIUM PLATE 0.0003-0.0005 THICK  
 REFER TO SOPM 20-42-05

ALL DIMENSIONS ARE IN INCHES

Bushing Detail  
 Figure 602

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REPAIR 4-1

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ILLUSTRATED PARTS LIST

1. This section lists and illustrates replaceable or repairable component parts. The Illustrated Parts Catalog contains a complete explanation of the Boeing part numbering system.

2. Indentures show parts relationships as follows:

Assembly

Detail Parts for Assembly

Subassembly

Attaching Parts for Subassembly

Detail Parts for Subassembly

Detail Installation Parts (Included only if installation parts may be returned to shop as part of assembly)

3. One use code letter (A, B, C, etc.) is assigned in the EFF CODE column for each variation of top assembly. All listed parts are used on all top assemblies except when limitations are shown by use code letter opposite individual part entries.

4. Letter suffixes (alpha-variants) are added to item numbers for optional parts, Service Bulletin modification parts, configuration differences (Except left- and right-hand parts), product improvement parts, and parts added between two sequential item numbers. The alpha-variant is not shown on illustrations when appearance and location of all variants of the part is the same.

5. Service Bulletin modifications are shown by the notations PRE SB XXXX and POST SB XXXX.

A. When a new top assembly part number is assigned by Service Bulletin, the notations appear at the top assembly level only. The configuration differences at detail part level are then shown by use code letter.

B. When the top assembly part number is not changed by the Service Bulletin, the notations appear at the detail part level.

6. Parts Interchangeability

Optional  
(OPT)

The parts are optional to and interchangeable with other parts having the same item number.

Supersedes, Superseded By  
(SUPSDS, SUPSD BY)

The part supersedes and is not interchangeable with the original part.

Replaces, Replaced By  
(REPLS, REPLD BY)

The part replaces and is interchangeable with, or is an alternate to, the original part.

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VENDORS

S0352 NIPPON MINIATURE BEARING CO LTD  
TOKYO, JAPAN

02758 NETWORKS ELECTRONIC CORP U S BEARING DIV  
9750 DE SOTO AVENUE  
CHATSWORTH, CALIFORNIA 91311-4409

09455 BFM TRANSPORT DYNAMICS CORP  
3131 WEST SEGERSTROM AVENUE PO BOX 1953  
SANTA ANA, CALIFORNIA 92702-1953

11815 CHERRY AEROSPACE FASTENERS DIV OF TEXTRON  
1224 EAST WARNER AVENUE PO BOX 2157  
SANTA ANA, CALIFORNIA 92707-0157

15653 KAYNAR TECHNOLOGY KAYNAR DIV  
800 SOUTH STATE COLLEGE BLVD PO BOX 3001  
FULLERTON, CALIFORNIA 92634-3001

15860 NEW HAMPSHIRE BALL BEARINGS, INCORPORATED ASTRO DIVISION  
155 LEXINGTON AVENUE  
LACONIA, NEW HAMPSHIRE 03246-2937

16746 SPECLINE INCORPORATED  
2230 MOUTON DR  
CARSON CITY, NV 89706

17446 HUCK MFG CO GOV CONTRACTS LOS ANGELES DIV SUB OF FED-MOGUL  
900 WATSON CENTER ROAD  
CARSON, CALIFORNIA 90745

21335 TORRINGTON CO FAFNIR BEARING DIV  
59 FIELD STREET  
TORRINGTON, CONNECTICUT 06790-4942

29666 HUCK MANUFACTURING CO SUB OF FEDERAL-MOGUL CORP  
6 THOMAS  
IRVINE, CALIFORNIA 92714

50294 NEW HAMPSHIRE BALL BEARINGS INC  
9730 INDEPENDENCE AVENUE PO BOX 2515  
CHATSWORTH, CALIFORNIA 91311-4323

52828 REPUBLIC FASTENER MFG CORP  
1300 RANCHO CONEJO BLVD  
NEWBURY PARK, CALIFORNIA 91320-1405

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VENDORS

56644 AURORA BEARING CO  
970 SOUTH LAKE STREET  
AURORA, ILLINOIS 60506-5929

56878 SPS TECHNOLOGIES INC AEROSPACE AND INDUSTRIAL PRODUCTS DIV  
HIGHLAND AVENUE  
JENKINTOWN, PENNSYLVANIA 19046

60119 MONADNOCK CO THE  
18301 ARENTH AVENUE PO BOX 1222  
CITY OF INDUSTRY, CALIFORNIA 91749

71087 BOOTS ACFT NUT DIV TOWNSEND CO SEE TEXTRON INC CHERRY  
FASTENER TOWNSEND DIV V11815

72962 HARVARD INDUSTRIES INC  
3 WERNER WAY SUITE 210  
LEBANON, NEW JERSEY 08833

73134 IMO INDUSTRIES INC HEIM BEARINGS DIV  
60 ROUND HILL ROAD PO BOX 430  
FAIRFIELD, CONNECTICUT 06430

80539 SPS TECHNOLOGIES INC AEROSPACE PRODUCTS DIV  
2701 SOUTH HARBOR BOULEVARD PO BOX 1259  
SANTA ANA, CALIFORNIA 92702-1259

81376 SOUTHWEST PRODUCTS COMPANY  
2240 BUENA VISTA STREET  
IRVINDALE, CALIFORNIA 91706

92215 FAIRCHILD IND INC FAIRCHILD AEROSPACE FASTENER DIV  
3010 W LOMITA BLVD  
TORRANCE, CALIFORNIA 90505-5102

97613 SARGENT CONTROLS & AEROSPACE/KAHR BEARING DIV  
5675 W BURLINGAME RD  
TUCSON, ARIZONA 85743

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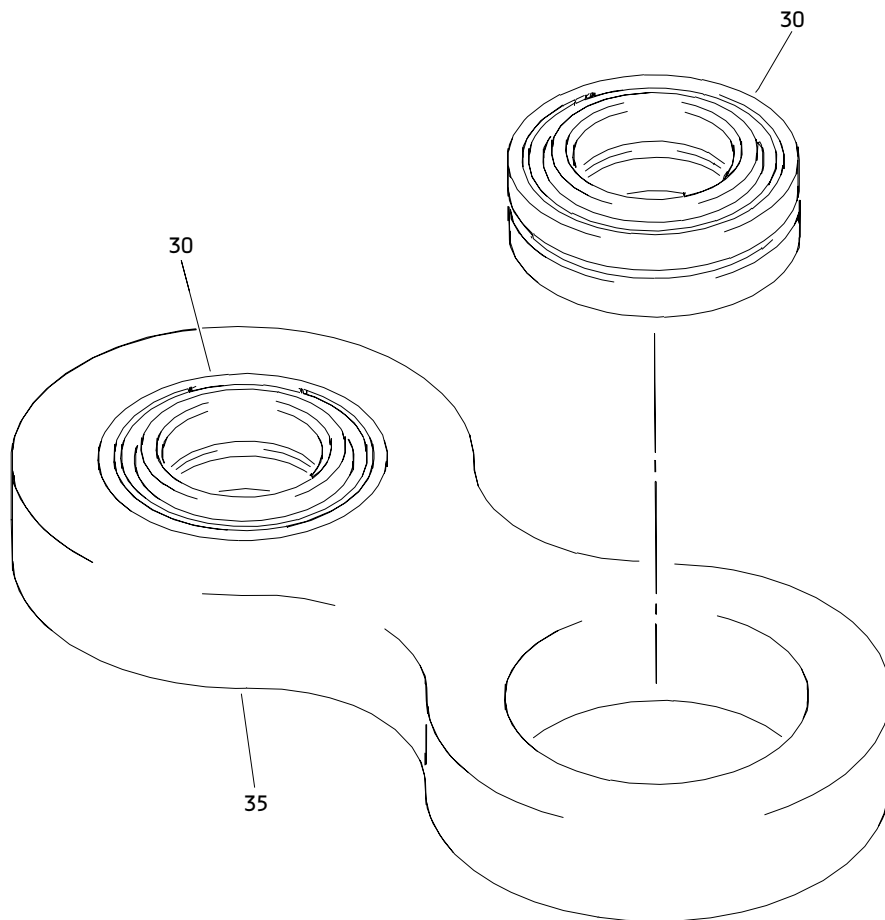
PART NUMBER	AIRLINE PART NO.	FIG.	ITEM	TTL REQ
ABG4V4		1	30	2
		1	30	2
AN960C416		2	10	2
BACB10CN4		1	30	2
BACB28X5M024		3	55	1
		6	25	1
BACB30GW6-61		3	30	3
BACB30NT3K34		6	5	2
BACB30NX6C5		3	5	3
BACC30K6		3	50	3
BACC30X6M		3	10	3
BACN10JC3		3	50A	3
BACN10JC3CD		6	15	2
BACN10JC4CM		2	15	2
BACN10JP4BCM		3	25	2
BACR15BA3AD3		3	20	4
BACR15BA4AD		3	60	2
BACW10CA3CCS		3	45	3
BACW10CA3CVS		3	40	3
BLN4-2230		1	30	2
BNG4F113		1	30	2
BRH10A3		3	50A	3
BRM100C4M		3	25	2
HG4-140		1	30	2
H10-3BAC		3	50A	3
KSBG4-57		1	30	2
K29646-3S		3	40	3
K29913-3S		3	45	3
MK2001-4BAC		3	25	2
MS14101-4		1	30A	2
MS20615-5MP		2	30	4
NAS1149D0332J		6	10	2
NAS623-3-61		3	30A	3
NAS6704-10		2	5	2
NBG4A		1	30	2
NC4G1		1	30	2
NS103198SE048		3	25	2
NS202101-02		3	50A	3
RMLH9075-3W		3	50A	3
RR4H11FS436		2	35	2
SALPYT6-61		3	30	3
T6S1032J		3	50A	3
T8081C428		3	25	2
VN201D1-048		3	25	2
VN303A02		3	50A	3
109A9207M4		3	25	2

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PART NUMBER	AIRLINE PART NO.	FIG.	ITEM	TTL REQ
2DCC6		3	50	3
251T1289-1		1	5	RF
251T1289-2		1	35	1
251T1507-1		2	45	1
251T1507-2		2	40	1
251T1508-1		2	50	1
251T1508-2		2	50A	1
251T1509-1		1	10	RF
		2	1	RF
251T1509-2		2	20	1
251T1509-3		2	25	1
251T1512-5		1	15	RF
		3	1	RF
251T1512-6		1	15A	RF
		6	1	RF
251T1552-4		3	15	1
251T1575-3		3	65	1
251T1575-5		3	70	1
251T1575-6		6	30	1
251T3741-26		3	35	3
251T3741-47		6	20	2
55768-4		1	30	2
70186-3S		3	40	3
70189-3S		3	45	3
922005-3		3	40	3
922006-3		3	45	3
942005-3		3	40	3
942006-3		3	45	3
96-02		3	50A	3

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Aileron Control Debris Guard Link Assembly  
Figure 1

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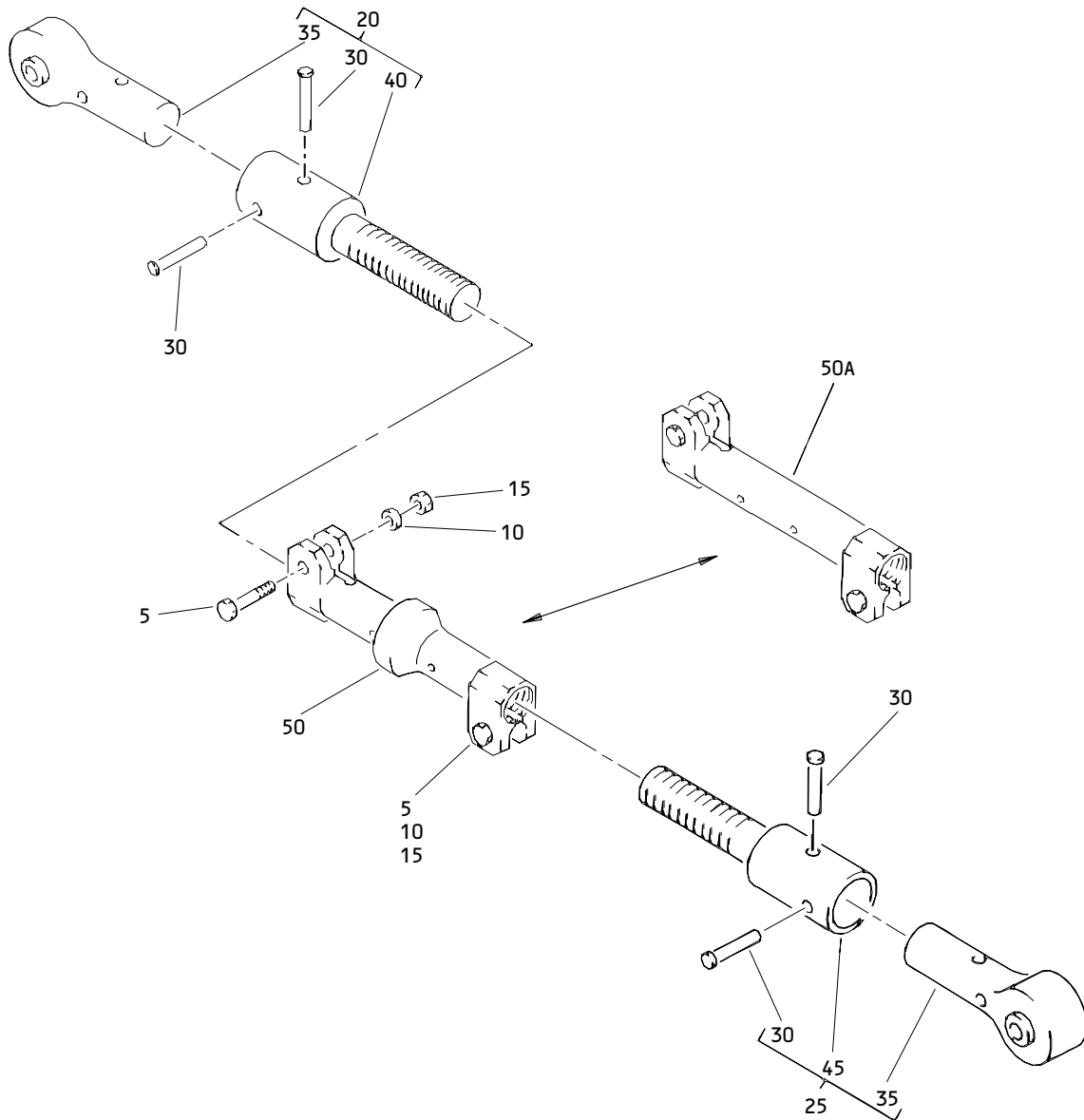
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FIG. & ITEM	PART NO.	AIRLINE PART NUMBER	NOMENCLATURE 1234567	EFF CODE	QTY PER ASSY
01-			COMPONENTS-AILERON CONTROL		
-5	251T1289-1		LINK ASSY-AIL. CONT DEBRIS GUARD	A	RF
-10	251T1509-1		ROD ASSY-AIL. CONT LATERAL CENTRAL CONT ACTR OUTPUT (FOR DETAILS SEE FIG. 2)	B	RF
-15	251T1512-5		FITTING ASSY-RH OVERRIDE MECH QUADRANT SPRT (FOR DETAILS SEE FIG. 3)	C	RF
R -15A	251T1512-6		FITTING ASSY-RH OVERRIDE MECH QUADRANT SPRT (FOR DETAILS SEE FIG. 6)	D	RF
-20	252T1200-5		DELETED		
-20A	252T1200-7		DELETED		
-25	252T1200-6		DELETED		
30	BNG4F113		.BEARING- (V16746) (SPEC BACB10CN4) (OPT BLN4-2230 (V81376)) (OPT HG4-140 (V02758)) (OPT KSBG4-57 (V97613)) (OPT 55768-4 (V09455)) (OPT NBG4A (V73134)) (OPT NC4G1 (V56644)) (OPT ABG4V4 (V50294)) (OPT AG4V25 (V15860)) (OPT ABG4V4 (VS0352)) (OPT ITEM 30A)	A	2
-30A	MS14101-4		.BEARING- (OPT ITEM 30)	A	2
35	251T1289-2		.LINK	A	1

- Item Not Illustrated

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Aileron Control Lateral Central Control Actuator Output Rod Assembly  
 Figure 2

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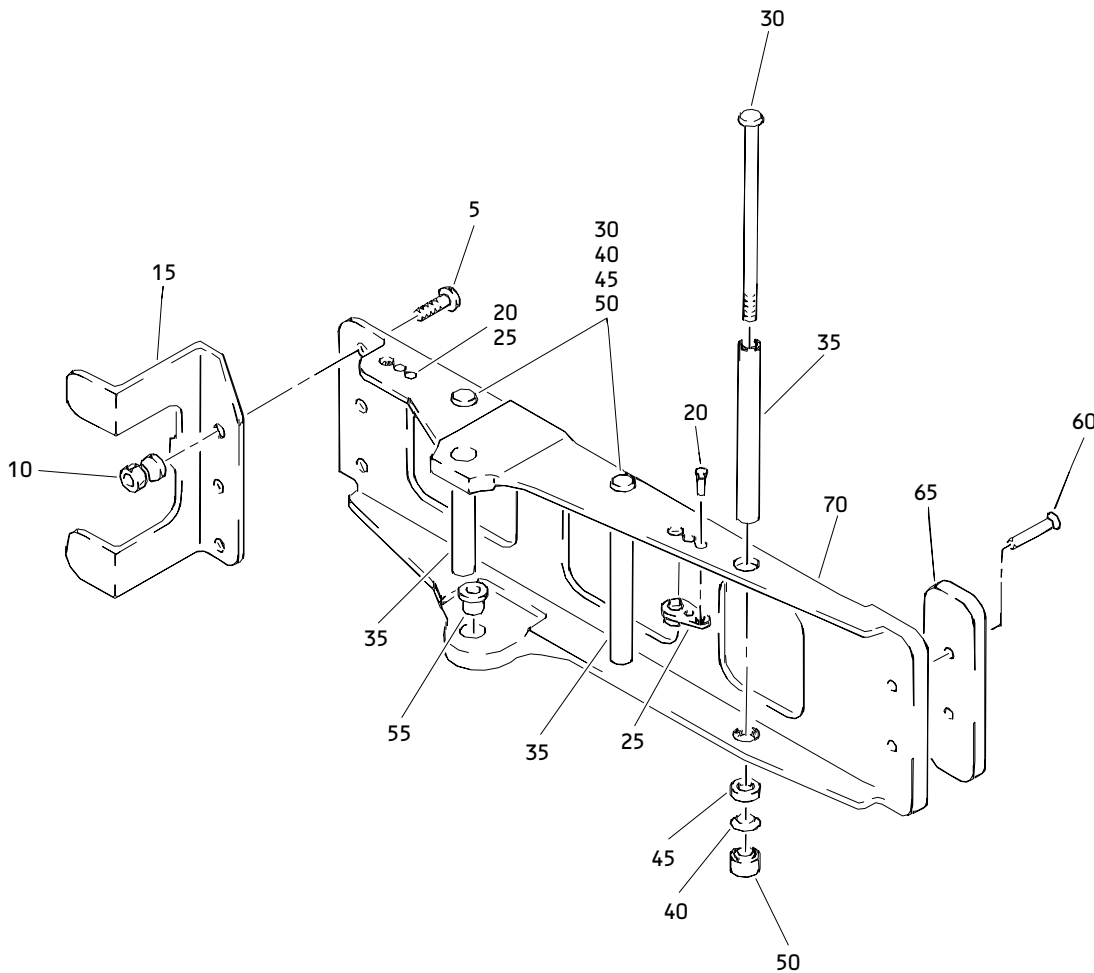
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FIG. & ITEM	PART NO.	AIRLINE PART NUMBER	NOMENCLATURE 1234567	EFF CODE	QTY PER ASSY
02- -1	251T1509-1		ROD ASSY-AIL. CONT LATERAL CENTRAL CONT ACTR OUTPUT	B	RF
5	NAS6704-10		.BOLT	B	2
10	AN960C416		.WASHER	B	2
15	BACN10JC4CM		.NUT	B	2
20	251T1509-2		.END ASSY	B	1
25	251T1509-3		.END ASSY	B	1
30	MS20615-5MP		..RIVET- (SIZE DETERMINE ON INST)	B	2
35	RR4H11FS436		..BEARING- (V21335)	B	1
40	251T1507-2		..FITTING- (USED ON ITEM 20)	B	1
45	251T1507-1		..FITTING- (USED ON ITEM 25)	B	1
50	251T1508-1		.BODY- (OPT ITEM 50A)	B	1
50A	251T1508-2		.BODY- (OPT ITEM 50)	B	1

- Item Not Illustrated

FIG. & ITEM	PART NO.	AIRLINE PART NUMBER	NOMENCLATURE 1234567	EFF CODE	QTY PER ASSY
03-					

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Right Hand Override Mechanism Quadrant Support Fitting Assembly  
Figure 3

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FIG. & ITEM	PART NO.	AIRLINE PART NUMBER	NOMENCLATURE 1234567	EFF CODE	QTY PER ASSY
03- -1	251T1512-5		FITTING ASSY-RH OVERRIDE MECH QUADRANT SPRT	C	RF
5	BACB30NX6C5		.BOLT	C	3
10	BACC30X6M		.COLLAR	C	3
15	251T1552-4		.BRACKET-UPR ZEE	C	1
20	BACR15BA3AD3		.RIVET	C	4
25	BRM100C4M		.NUTPLATE- (V52828) (SPEC BACN10JP4BCM) (OPT MK2001-4BAC (V15653)) (OPT NS103198SE048 (V80539)) (OPT VN201D1-048 (V92215)) (OPT 109A9207M4 (V72962)) (OPT T8081C428 (V11815))	C	2
30	SALPYT6-61		.BOLT- (V11815) (SPEC BACB30GW6-61) (OPT SALPYT6-61 (V29666)) (OPT SALPYT6-61 (V17446)) (OPT ITEM 30A USED WITH ITEM 50A) (USED WITH ITEM 50)	C	3
-30A	NAS623-3-61		.BOLT- (OPT ITEM 30 USED WITH ITEM 50) (USED WITH ITEM 50A)	C	3

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FIG. & ITEM	PART NO.	AIRLINE PART NUMBER	NOMENCLATURE 1234567	EFF CODE	QTY PER ASSY
03-35	251T3741-26		.SPACER	C	3
40	K29646-3S		.WASHER- (V15653) (SPEC BACW10CA3CVS) (OPT 70186-3S (V56878)) (OPT 922005-3 (V60119)) (OPT 942005-3 (V60119))	C	3
45	K29913-3S		.WASHER- (V15653) (SPEC BACW10CA3CCS) (OPT 70189-3S (V56878)) (OPT 922006-3 (V60119)) (OPT 942006-3 (V60119))	C	3
50	2DCC6		.COLLAR- (V11815) (SPEC BACC30K6) (OPT 2DCC6 (V17446)) (USED WITH ITEM 30)	C	3
-50A	H10-3BAC		.NUT- (V15653) (SPEC BACN10JC3) (OPT NS202101-02 (V80539)) (OPT RMLH9075-3W (V72962)) (OPT T6S1032J (V71087)) (OPT VN303A02 (V92215)) (OPT 96-02 (V80539)) (OPT BRH10A3 (V52828)) (USED WITH ITEM 30A)	C	3

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FIG. & ITEM	PART NO.	AIRLINE PART NUMBER	NOMENCLATURE	EFF CODE	QTY PER ASSY
			1234567		
03-55	BACB28X5M024		.BUSHING	C	1
60	BACR15BA4AD		.RIVET- (SIZE DETERMINE ON INST)	C	2
65	251T1575-3		.SPACER	C	1
70	251T1575-5		.FITTING	C	1

- Item Not Illustrated

FIG. & ITEM	PART NO.	AIRLINE PART NUMBER	NOMENCLATURE	EFF CODE	QTY PER ASSY
			1234567		
04-1	252T1200-5		DELETED		

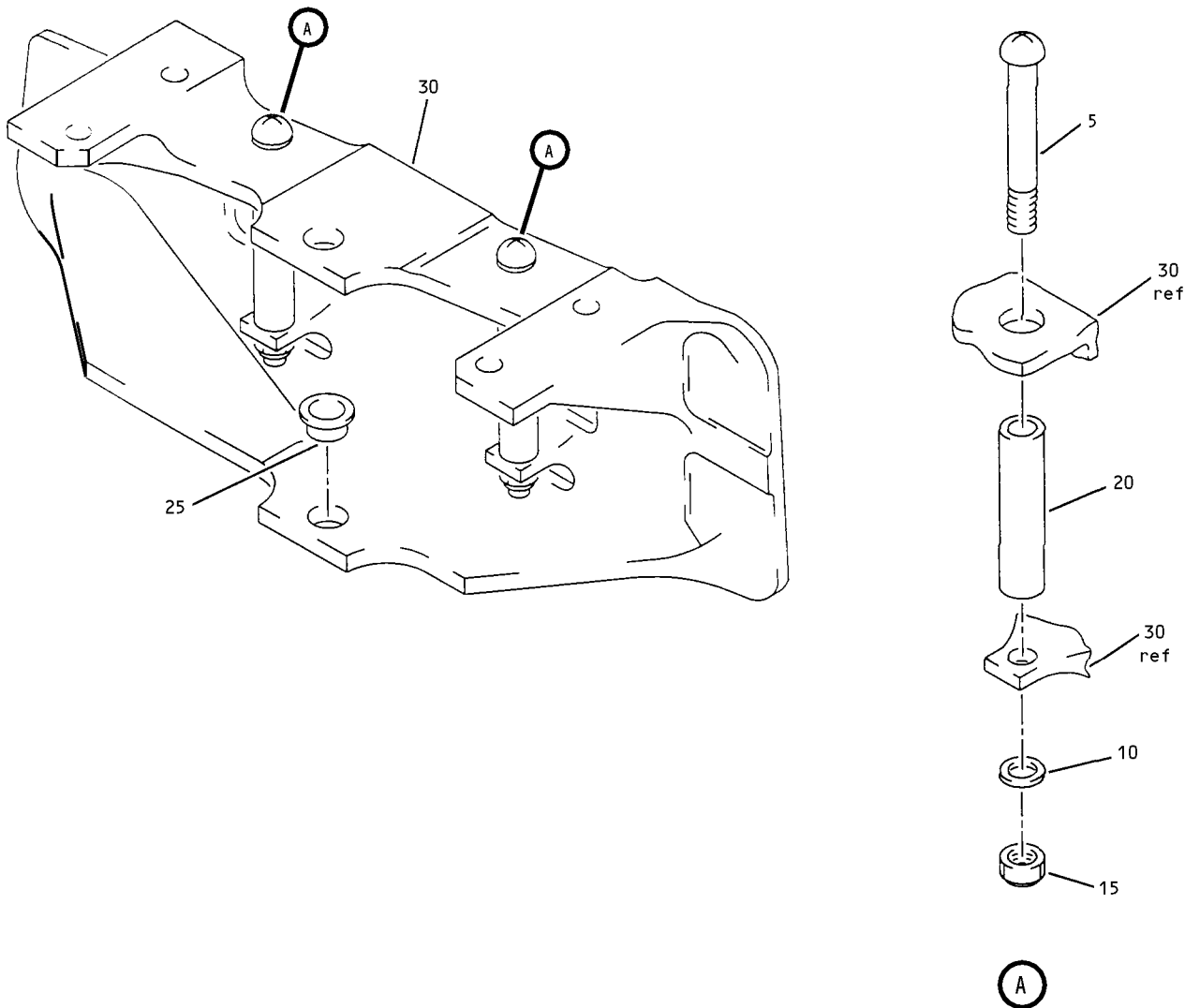
- Item Not Illustrated

FIG. & ITEM	PART NO.	AIRLINE PART NUMBER	NOMENCLATURE	EFF CODE	QTY PER ASSY
			1234567		
05-1	252T1200-6		DELETED		

- Item Not Illustrated

FIG. & ITEM	PART NO.	AIRLINE PART NUMBER	NOMENCLATURE	EFF CODE	QTY PER ASSY
			1234567		
06-					

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Right Hand Override Mechanism Quadrant Support Fitting Assembly  
 Figure 6

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FIG. & ITEM	PART NO.	AIRLINE PART NUMBER	NOMENCLATURE 1234567	EFF CODE	QTY PER ASSY
R 06- -1	251T1512-6		FITTING ASSY-RH OVERRIDE MECH QUADRANT SPRT	D	RF
R 5	BACB30NT3K34		.BOLT	D	2
R 10	NAS1149D0332J		.WASHER	D	2
R 15	BACN10JC3CD		.NUT	D	2
R 20	251T3741-47		.SPACER	D	2
R 25	BACB28X5M024		.BUSHING	D	1
R 30	251T1575-6		.FITTING	D	1

- Item Not Illustrated

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